## Effect of thermal pressure in converging detonation waves

#### V P SINGH

Terminal Ballistics Research Laboratory, Sector-30, Chandigarh 160020, India

MS received 19 January 1984; revised 21 September 1984

Abstract. Propagation of converging detonation waves in various explosives is studied using the equation of state, which considers both the thermal and elastic pressures. It is seen that the rate of increase of thermal pressure is higher than that of the elastic pressure during convergence. The present equation of state is better since it also gives the variation of temperature, whereas the polytropic form of the equation of state is independent of temperature. It is seen that the total detonation pressure is slightly greater than the elastic pressure. Results are compared with those reported elsewhere.

Keywords. Converging detonations; thermal pressures in detonation.

PACS No. 47-40

#### 1. Introduction

Convergence of detonation waves in solid explosives is of immense importance, when very high pressures and temperatures are required. The problem of converging detonation waves in solid and gaseous explosives was studied by various authors (Abarbanel 1967; Lee 1967; Singh 1978, 1983), using the polytropic equation of state. Mader (1979) used BKW equation of state and studied the convergence of detonation waves in solid explosives by numerical methods. It was shown by Singh (1981) that BKW equation of state gives much lower pressures than that given by the polytropic equation of state (Singh 1978, hereafter to be referred as paper I). It is known that explosive products are not purely solids but have dual character—that of solids as well as gases. For such types of media, Hornberg (1978) suggested an equation of state, where thermal as well as elastic pressures were considered. Such an equation was first suggested by Zeldovich and Kompaneets (1960) which Hornberg called Zeldovich's generalised equation of state of solid explosives.

We have studied the convergence of detonation waves in solid explosives using Zeldovich's generalised equation of state. Jump conditions for over-driven detonation waves are derived in § 2. Variations of pressure and temperature as the detonation wave converges are studied in § 3. The results are discussed in § 4.

#### 2. Formulation of the problem

Let us assume that a spherical detonation wave travels from the surface of a solid explosive charge, towards its centre. At any time t, let R be the position of the detonation front, where R is measured from the centre of convergence. Let p, p, U and u

527

be the pressure, density, over-driven detonation velocity and particle velocity of detonation products respectively. The quantities behind the detonation front are related with those ahead of it by

$$\rho(U - u) = \rho_0 U, \quad (1)$$

$$\rho(U - u)^{2} + p = \rho_{0}U^{2}, \qquad (2)$$

$$\frac{1}{2}(U-u)^2 + E + \frac{p}{\rho} = \frac{1}{2}U^2 + Q,$$
 (3)

where E is the internal energy and  $\rho_0$  is the density of the unexploded charge. It is assumed that the heat of detonation Q remains constant during the process of convergence, and is given by (Hornberg 1978)

$$Q = \underbrace{\left[n + (1 + \Gamma)y\right]}_{\Gamma(x+1)} \underbrace{\left[\frac{1}{(n-1)} + \frac{y}{\Gamma} - \frac{(1+y)^2}{2\left[n + (1+\Gamma)y\right]}\right]}_{\Gamma(x+1)} D^2, \tag{4}$$

where Q is measured per unit mass and y is a function of temperature T and is given by

$$y = \left[\frac{p/\rho}{c_v \Gamma T} - 1\right]. \tag{5}$$

In the above relation  $c_v$ ,  $\Gamma$  and T are respectively the specific heat at constant volume, the Grüneisen constant and the temperature of detonation products, y is defined as the ratio of thermal pressure to elastic pressure (Hornberg 1978)

$$y = \frac{c_n \Gamma v^{n-1} T}{\tilde{C}},$$
(6)

where  $\tilde{C}$  is defined in (7). Total pressure of detonation products is defined as the sum of elastic and thermal pressures and is given as follows

$$p = \tilde{C} \hat{\rho}^* + \frac{c_v \Gamma T}{v}, \qquad (7)$$

where the first term gives the elastic pressure and the second term the thermal pressure and v is inverse of density  $\rho$ . Internal energy E is given by

$$E = \frac{p}{(n-1)\rho} \left[ 1 + \frac{(n-1)}{\Gamma} y \right]. \tag{8}$$

Equations (1)–(8) show that there are seven unknown quantities p, U, u,  $\rho$ , E, y and T, to solve which we have six equations (1)–(3), (5), (6) and (8). Solving these equations using (4) and taking z and y as independent parameters, one gets

$$u = \frac{(1+\overline{y})DF}{(n+1)\overline{f_1}}$$
(9)

$$U = Dz/F$$
 (10)

$$\rho/\rho_0 = \frac{(n+1)\overline{f_1}z}{n\overline{f_4}G^2z_0},$$
(11)

where

$$z = \frac{(n+1)\overline{f_1}p}{(1+\overline{y})\rho_0 D^2},$$

$$f_1 = \left[1 + \left(\frac{2+\Gamma}{n+1}\right)y\right]$$

$$f_2 = \left[1 + \frac{(n-1)(2+\Gamma)}{\Gamma(n+1)}y\right]$$

$$f_3 = \left[1 + \frac{(n-1)}{\Gamma}y\right]$$

$$f_4 = \left[1 + \frac{(1+\Gamma)}{n}y\right].$$

$$F = \left[1 + \frac{2f_3}{(n+1)f_2}\left\{\frac{(n+1)\overline{f_1}}{(1+\overline{y})}(z-z_0) - 1 + z_0'\right\}\right]^{1/2}$$

$$G = \left[1 + \frac{(n-1)(1+y)\overline{f_1}(z-z_0)}{nf_2\overline{f_4}z_0} + \frac{(1+y)(1+\overline{y})(n-1)}{n(n+1)f_2\overline{f_4}}\left(\frac{z_0-1}{z_0}\right)\right]^{1/2}$$

$$z_0 = \left[\frac{(1+y)\overline{f_3}}{(1+\overline{y})f_3}\right]$$
(13)

The sound velocity c is given by

$$c = \frac{n}{(n+1)} \frac{DG\overline{f_4}}{\overline{f_1}} \left( \frac{\overline{f_3}f_4}{\overline{f_3}\overline{f_4}} \right)^{1/2}.$$
 (14)

In relation (9)–(13), the bar above the symbols denotes the value of the functions at the Chapman-Jouguet (abreviated as C-J plane) surface. At the C-J plane z=1 and (1)–(3) reduce to (21) of Hornberg (1978) for non-overdriven detonation waves.

Equations (9)–(11) are the generalised form of jump conditions for the equation of state (7). If we put y = 0 these equations reduce to equations (1)–(4) of paper I.

Equations (9) to (11) relate U, u and  $\rho/\rho_0$  with y and z which are still unknown. It is our aim to express them as a function of the radius of convergence. Thus we require two extra relations to express all detonation parameters in terms of the radius of convergence R. Once y is determined, temperature T can be evaluated from (6). One extra relation is obtained by the Whitham's method of characteristics (Whitham 1958), which is discussed in § 3.

#### 3. Variation of pressure with the ratio of convergence

Following paper I, we use the equation of motion along the positive characteristic axis as the extra relation relating detonation parameters. The characteristic form of the

equation of motion is

$$dp + \rho u du + \frac{2\rho c^2 u}{(u+c)} \frac{dR}{R} = 0$$
 (15)

where c is the sound velocity behind the detonation front. Substituting for p, p, c and u from (9)–(14) in (15), one gets after simplifications,

$$H_1(dy/dR) + H_2(dz/dR) + H_3 = 0$$
 (16)

where

$$H_{1} = \frac{z}{FGz_{0}} \left( \frac{\overline{f}_{3}f_{4}}{f_{3}\overline{f}_{4}} \right)^{1/2} \left( \frac{n-1-\Gamma}{\Gamma} \right) \left( \frac{n-1}{\Gamma} \right) \left[ \frac{2(z-z_{0})(n-1)\overline{f}_{1}}{(n+1)(1+\overline{y})f_{2}^{2}} - \frac{2(n-1)(1-z_{0})}{(n+1)^{2}f_{2}^{2}} + \frac{2n\overline{f}_{3}\overline{f}_{4}}{(n+1)(1+\overline{y})f_{2}f_{3}} \right]$$

$$(17)$$

$$H_2 = \left[1 + y + \frac{\overline{f}_1 f_3 z}{f_2 F G z_0} \left(\frac{\overline{f}_3 f_4}{f_3 \overline{f}_4}\right)^{1/2}\right]$$
(18)

$$H_3 = \frac{2nz}{R} \left[ \frac{(1+y)F f_4}{(1+\sqrt{y})F + nG f_4 (f_3 f_4/f_3 \overline{f_4})^{1/2}} \right]$$
(19)

Equation (16) is a non-linear differential equation in y and z. In order to solve this equation, we require one more relation in dy/dR and dz/dR which is obtained from (6) and (7) with the help of (11) as

$$G_1(dz/dR) = G_2(dy/dR) \qquad (20)$$

where

$$G_1 = \left[ \frac{n-1}{z} - n \left( \frac{G^2 - 1}{G^2} \right) \overline{f_1} \right], \tag{21}$$

$$G_2 = \left\lceil \left(\frac{n-1}{1+y}\right) - \frac{n(n-1)}{\Gamma f_3} + n \left(\frac{G^2-1}{G^2}\right) \left\{ \left(\frac{n-1}{\Gamma}\right) \left(\frac{n-1-\Gamma}{(n+1)f_2f_3}\right) \right\} \right\rceil$$

$$+\frac{nf_4z_0(n-1-\Gamma)}{(n+1)\Gamma(1+y)f_3f_5}$$
 (22)

Function  $f_5$  in (21) and (22) is given as

$$f_5 = \frac{(1-z_0)n\overline{f_4}}{(n+1)} \left[1 + \frac{(z-1)(1+\overline{y})}{n\overline{f_4}(1-z_0)}\right]. \quad (23)$$

Solving (16) and (20) for dz/dR, dy/dR we get

$$dv/dR = -H_1G_1/[H_1G_1 + H_2G_2],$$
 (24)

$$dz/dR = -H_1G_2/[H_1G_1 + H_2G_2].$$
 (25)

Equations (24) and (25) are two differential equations which give the variation of y and z in terms of the radius of convergence R, T is obtained from (6). Once z and y are known, other parameters can be obtained from the jump conditions (9)–(11).

# 4. Discussion of results and conclusions

Equations (24) and (25) are integrated by the Runge-Kutta method of fourth order, using a computer DEC-20. Variations of detonation parameters are calculated for four basic explosives and one mixture *i.e.* PETN, RDX, TETRYL, TNT and composition-B (RDX:TNT; 60:40). To calculate y,  $c_v$  and values of  $\Gamma$  for the detonation products of the explosives are required. We have calculated  $c_v$  using input data given by Mayer (1981). Value of  $c_v$  and  $\Gamma$  are given in table 1. To evaluate  $\Gamma$  we have taken ( $\partial \ln D/\partial \ln \rho_0$ ) = 0.81 as given by Zeldovich and Kompaneets (1960).

Figure 3 gives a plot of the variation of y and T with  $(R/R_0)$ , where  $R_0$  is the total radius of the charge. It is seen that y first decreases and then increases, i.e. initially the rate of increase of elastic pressure is greater as compared to the thermal pressure but later on in few cases (e.g. TNT) the thermal pressure becomes greater. Variation of y i.e. ratio of thermal pressure to elastic pressure for five explosives are given in table 2 and the total pressure is given in table 3. From figure 4, it is seen that during convergence, the thermal pressure increases at a higher rate compared to the elastic pressure. The

Table 1. Values of the characteristic detonation parameters of high explosives.

explosives.							
Explosive	ρ <sub>0</sub> (g/oc)	D (m/s)	и	г	c, (cal/g-	y degree)	$\hat{C} \times 10^{10}$
RDX Tetryl PETN Comp-B	1-64 1-64 1-65 1-69 1-64	8475 7556 7854 7862 6950	3-02 2-97 3-04 2-75 3-16	0-5603 0-4950 0-5862 0-1985 0-7396	0-3654 0-3755 0-3331 0-3746 0-3913	0-2126 0-2364 0-2381 0-0784 0-3756	2-77 2-29 2-32 2-80 1-66

Table 2. Variation of y vs  $R/R_0$  for various explosives.

Distance R/R <sub>0</sub>	TNT	PETN	Tetryl	Comp-B	RDX
1-0	0-3757	0-2381	0-2362	0-0784	0-2126
0-9	0-3653	0-2314	0-2295	0-0759	0-2067
0-8	0-3658	0-2251	0-2228	0-0735	0-2012
0-7	0-3470	0-2198	0-2169	0-0713	0-1967
0-6	0-3413	0-2167	0-2121	0-0695	0-1938
0-5	0-3402	0-2166	0-2100	0-0686	0-1943
0-4	0-3478	0-2229	0-2130	0-0693	0-2005
0-3	0-3740	0-2419	0-2264	0-0737	0-2186
0-2	0-4487	0-2949	0-2665	0-0870	0-2684
0-1	0-7419	0-4879	0-4163	0-1322	0-4479
0-09	0-8181	0-5356	0-4516	0-1422	0-4921
0-08	0-9223	0-5974	0-4972	0-1547	0-5477
0-07	1-0377	0-6803	0-5570	0-1700	0-6227

Table 3. Variation of pressure p (kbar) vs  $R/R_0$  for various explosives.

Distance R/R <sub>6</sub>	TNT	PETN	Tetryl	Comp-H	RDX
1.0	210-646	290-576	253-873	287-203	312-943
0.9	227-498	292-316	274-183	310-176	337-978
0.8	248-562	320-736	299-570	338-896	370-837
0.7	278-053	358-629	332-574	376-232	413-084
0-6	315-969	405-995	378-271	427-929	475-673
0.5	368-691	476-368	441-739	499-729	550-779
0-4	452-889	584-633	543-288	608-865	675-956
0-3	598-230	768-684	710-845	801-289	888-757
0-2	912-097	1171-970	1068-810	1200-500	1364-430
0-1	2089-610	2620-020	2325-480	2492-900	3029-280
0-09	2411-900	3004-360	2640-280	2803-070	3473-660
0.08	2864-780	3518-620	3059-170	3202-280	4052-260
0.07	3380-870	4235-880	3630-380	3719-240	4866-260

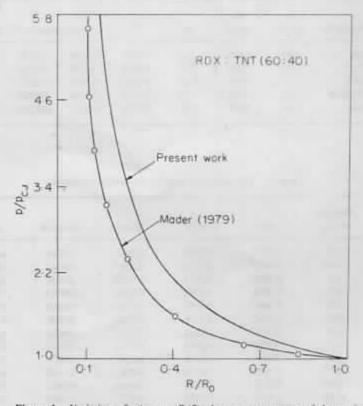


Figure 1. Variation of  $p/p_{CJ}$  vs  $R/R_0$  due to convergence of detonation in a typical explosive (composition-B).

pressure when the polytropic equation of state is used (paper I), is initially lower but becomes higher than the total pressure of the present paper, as the detonation wave moves towards the centre of convergence. A typical case of composition-B is shown in figure 2.

Figures 1 and 2 give the variation of  $p/p_{CI}$  and U/D vs  $(R/R_0)$  as the detonation wave converges from its initial radius  $R_0$  to the centre of convergence. Results are compared with that of Mader (1979). The pressure and detonation velocity obtained by present method is slightly greater as compared to that of Mader (1979). This is because we have used a different equation of state. Mader has used the BKW equation of state with Arrhenius rate law. It is to be noted that BKW equation of state is calibrated for CJ values of detonation parameters and its accuracy at higher pressures is under question. Moreover it has been shown (Singh 1981) that the BKW equation of state gives much lower pressures as compared to that given by polytropic equation of state. This lower trend of Mader's values may be due to BKW equation of state only. We have not come across any experimental data, by which the law of convergence in explosives may be verified.

The Zeldovich equation of state is more realistic as it reduces to other forms of equations of state in special cases. The equation reduces to

$$pv = RT$$

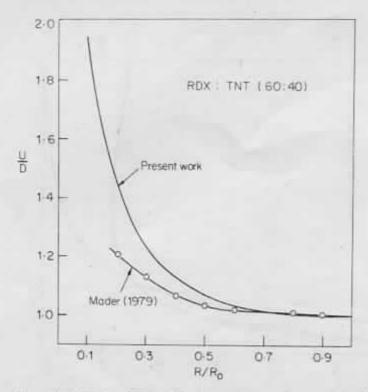


Figure 2. Variation of U/D vs  $R/R_0$  due to convergence in composition-B.

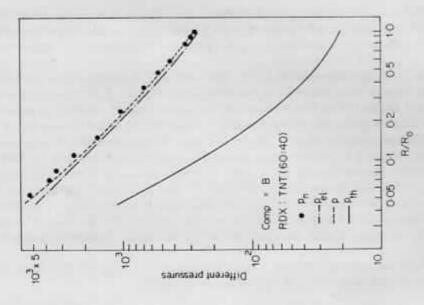


Figure 4. Variation of different pressures due to convergence in composition-B.

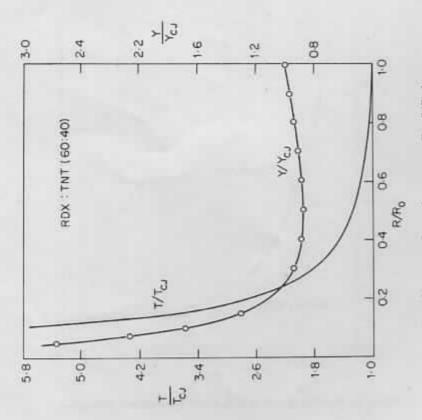


Figure 3. Variation of pressure ratio y and temperature T vs  $R/R_0$  due to convergence in composition-B.

for  $y \to \infty$  and  $p = \tilde{C} \rho^n$  for y = 0. This means, it represents both gaseous and solid behaviour of detonation products. The only approximation in the present paper is regarding heat of detonation, which is assumed to be constant during the process of convergence.

### Acknowledgements

Thanks are due to Shri D S Murty and Col. Harindar Singh for their interest and Shri J P Sirpal, Director for giving permission to publish the paper.

#### References

Abarbanel S 1967 Israel. J. Technol. 5 235
Hornberg H 1978 Propellants Explos. 3 97
Lee B H K 1967 AIAA. J. 5 1997
Mader C L 1979 Numerical modelling of detonations (Los Angeles: University California Press)
Mayer R 1981 Explosives Verlag Chemie, Wheinheem
Singh V P 1978 J. Indian Math. Soc. 42 381
Singh V P 1981 Proc. Workshop on Terminal Ballistics, IAT, Pune 103
Singh V P 1983 Def. Sci. J. 33 53
Whitham G B 1958 J. Fluid Mech. 4 337
Zeldovich J B and Kompaneets A S 1960 Theory of detonations (New York: Academic Press)